

# Chapter 5: Nonlinear Finite Element for Solids

## 5.1 Introduction

T

2D 3D . M , . E  
. I , . H ,  
\_\_\_\_\_ . T

M

\_\_\_\_\_ / \_\_\_\_\_ . A  
. F ,

\_\_\_\_\_

M US

V \_\_\_\_\_

I \_\_\_\_\_ ,  
 . I \_\_\_\_\_ ,  
 . I \_\_\_\_\_

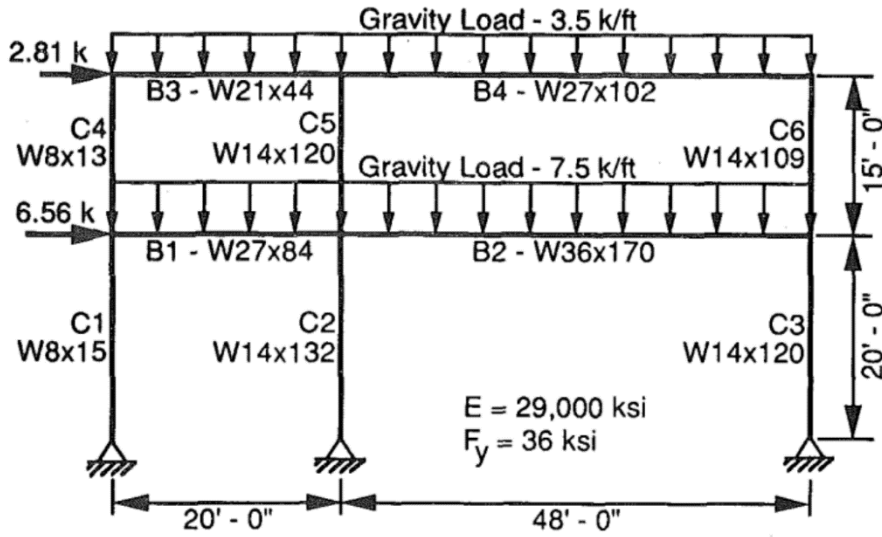
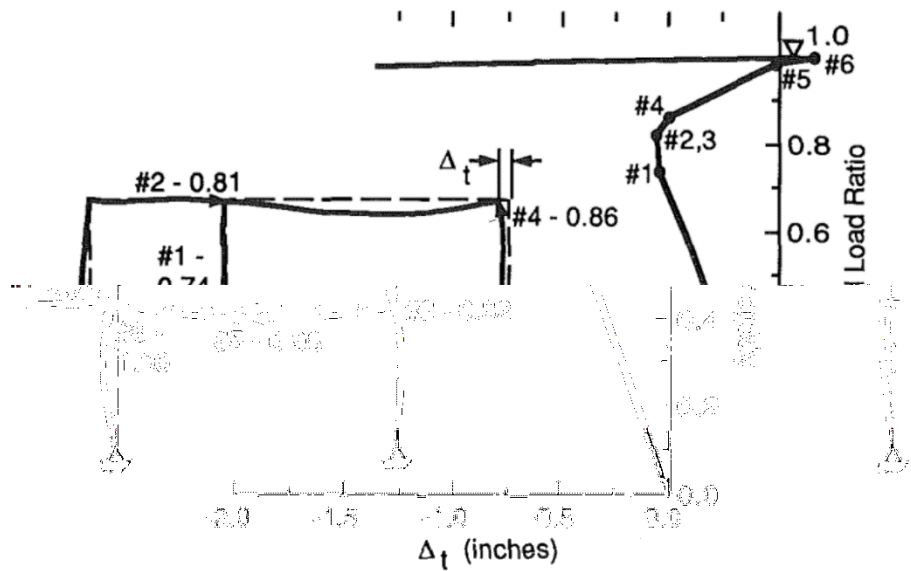


FIG. 5. Loads and Dimensions for Frame U-P36H



P \_\_\_\_\_

## 5.2 Nonlinear Phenomena

M

- **Geometrical nonlinearity**

*the strains are still small. G*

- **Finite deformations**

. H

- **Material nonlinearity: M**

. A

material

. T

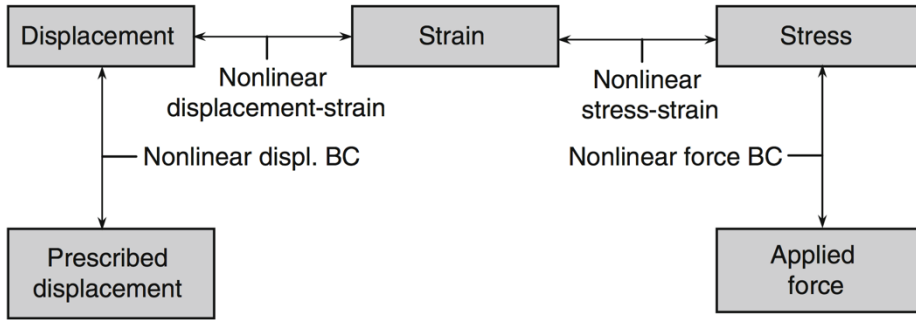
- **Stability problems**

. G includes bifurcations like buckling of frames or shells but can also be connected to limit points which indicate snap-through behaviour of a structure. Material instabilities come along with necking or shear bands in metals but also geo-materials. The origin of these lies either in an instability of the equilibrium equations or in the loss of positive definiteness related to the incremental constitutive tensor of the material. Both instabilities react in a very sensitive way to imperfections.

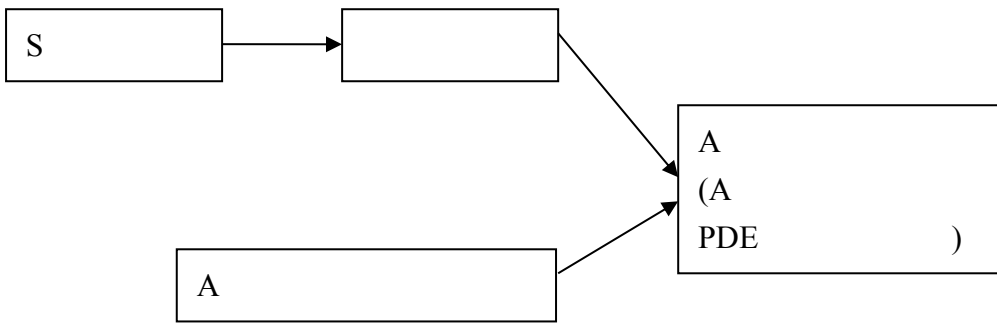
- **Contact problems** are characterized by stemming from the boundary are associated with **contact** between two bodies or deformation dependent loading.

- **Coupled problems** occur when different interacting fields which describe e.g. solids, heat in solids or fluids are needed to formulate a complex physical problem. Examples are thermomechanical coupling, fluid-structure-interaction or problems in which chemical reactions, heat generation and conduction and mechanical stresses have to be coupled to model an engineering process, like the design of a new material. In all cases, nonlinearities in each of the different field equations have to be considered.

F



P



T , ( E . (2.34) (2.38)):

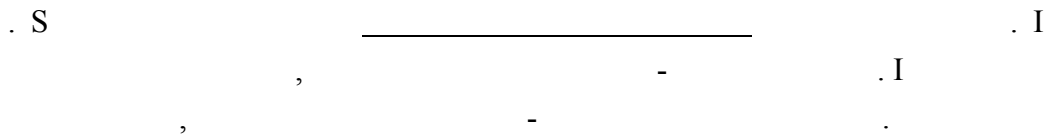
$$(2.34) \quad \mathbf{w}^T (\mathbf{K}\mathbf{d} - \mathbf{f}) = 0$$

$$(2.38) \quad \mathbf{K}_F \mathbf{d}_F = \mathbf{f}_F - \mathbf{K}_{EF}^T \bar{\mathbf{d}}_E \Rightarrow \mathbf{d}_F = (\mathbf{K}_F)^{-1} (\mathbf{f}_F - \mathbf{K}_{EF}^T \bar{\mathbf{d}}_E)$$

T ( ) K f K  
 d. **cannot** d K  
 f . A d K  
 f  $\mathbf{K}_F \mathbf{d}_F$   $\mathbf{f}_F - \mathbf{K}_{EF}^T \bar{\mathbf{d}}_E$ .

nonlinear algebraic equations.

I



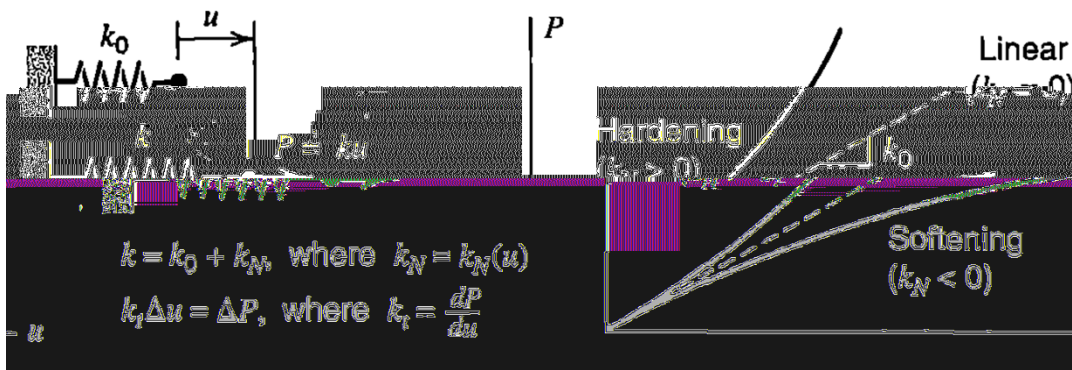
5.3 Solution Methods for Nonlinear Algebraic Equations

$$\mathbf{K}_F \mathbf{d}_F = \mathbf{f}_F \quad (\bar{\mathbf{d}}_E = \mathbf{0})$$

I

$$(5.1) \quad g(u) = u = u(x)$$

u . A



T P u

$$(5.2) \quad ku = P \quad (k_0 + k_N)u = P \quad k_N = k_N(u)$$

$$\mathbf{K}_F \mathbf{d}_F = \mathbf{f}_F, \quad k, \quad ku, \quad u, \quad P, \quad u$$

C **tangent stiffness,**

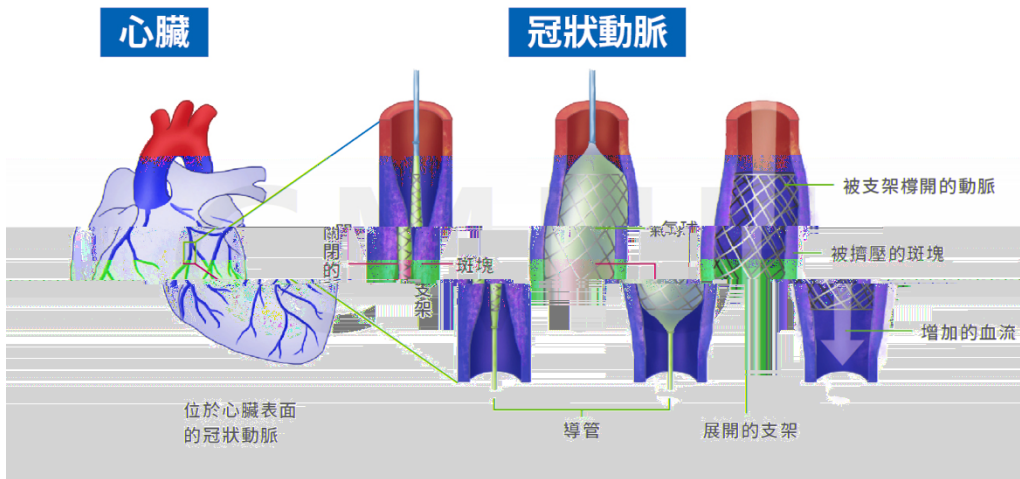
$$(5.3) \quad k_t = \frac{dP}{du} = k + \frac{\partial k_N}{\partial u} u$$

$$\underline{\quad} \quad P \quad u \quad .$$

Remarks:

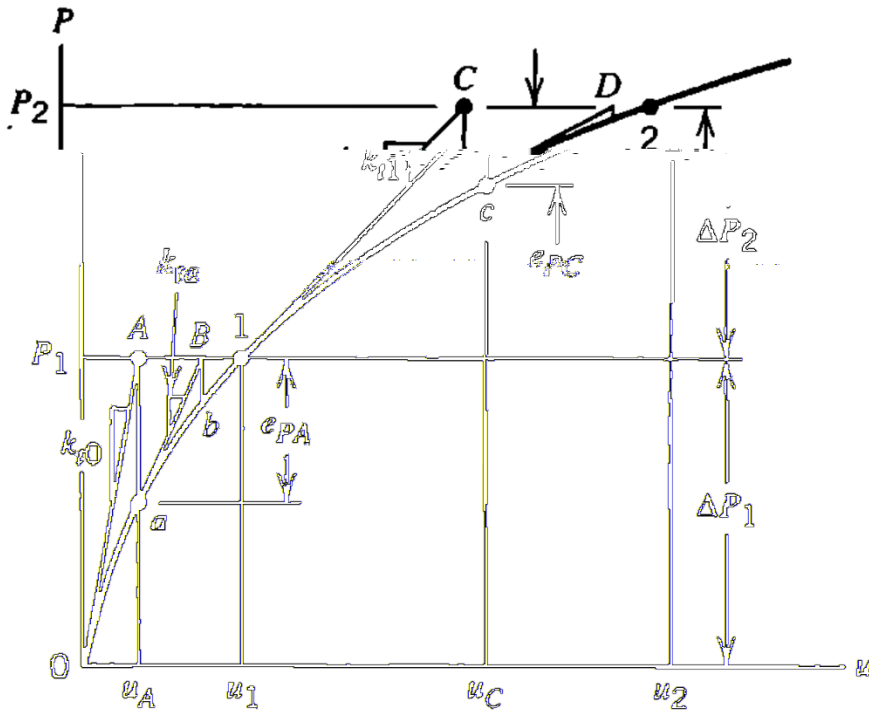
1. N tangent stiffness  $k_t$  NOT  $k ($   
 , tangent stiffness matrix  $K_T$   
 $K_F)$

2. I - ,  $u, I$   
 $K_F$   $f_F$   $P,$   $d_F \cdot E$  ,  
 ( . , \_\_\_\_\_ :  
 , , ).



pa gure 1.2: 1988 Aloha Airlines accident—the shaded area illustrates the fuselage lost at cruising altitude.

5.3.1 Newton-Raphson Method for Single DOF



I , N ' ,  
 . H P  
 u , \_\_\_\_\_

I E . (5.2),  $u = 0$  . T  $P_1$   
 $u_1$  .

T  $k_{t0}$  ,  
 ;  $\Delta P = P_1$   
 :

$$(5.4) \quad k_{t0} \Delta u = \Delta P_1 \quad \Delta u = (k_{t0})^{-1} \Delta P_1 \quad u_A = 0 + \Delta u$$

H  $u_A$   $u_1$  . T \_\_\_\_\_  
 $P_1$  .

Q: ( )  $e_{PA}$  ?

A:

(5.5)

Remark:

$ku_A$

\_\_\_\_\_

**equilibrium iterations**

( ) .

$P_1$

**a**

**tangent**

a. T

$u_B$ :

$$(5.6) \quad k_{ta} \Delta u = e_{PA} \quad \Delta u = (k_{ta})^{-1} e_{PA} \quad u_B = u_A + \Delta u$$

T

$P_1$ .

Q:

( )  $e_{PB}$ ?

A:

(5.7)

T

**b**

$\Delta u$

$u_B + \Delta u$  . A

$\Delta u$

$u_1$ .

A

$\Delta P_2$

$P$  . E

E . (5.4)

(5.5)

$$(5.8) \quad \Delta u = (k_t)^{-1} \Delta P \quad u_C = u_1 + \Delta u$$

$$(5.9) \quad e_{PC} = P_2 - ku_C$$

$$k = k(u)$$

$u_C$ .

C

1, 2, 3, ...

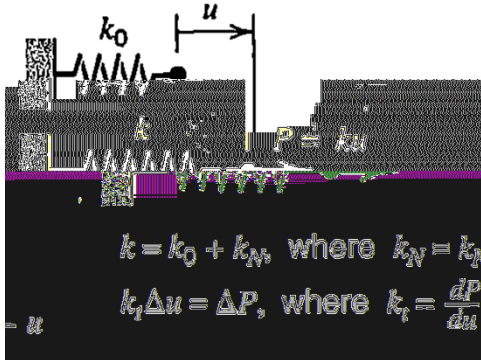
P

u

T

Remark: N -R :  
 prediction ( . ., E . (5.8)) *the tangent stiffness*  
 . T correction ( . ., E . (5.9))

E \_\_\_\_\_: N -R  
 :  $k_0 = 50, k_N = 5u, P = 100.$



(Solution)

T :  
 $k = k_0 + k_N = 50 - 5u$

T :  
 $P = ku = (50 - 5u)u$

Q: ?  
 A:

L  $P_1 = P = 100$ :

I \_\_\_\_\_ 1

Prediction (E . (5.4)):

$$k_{i0} = 50$$

$$k_{i0} \Delta u = 100 \quad \Delta u = \frac{1}{50} 100 \quad u_A = 0 + \Delta u = 2.0$$

Correction (E . (5.5)):

$$k = 50 - 5u_A = 40$$

$$e_{PA} = P_1 - ku_A = 20$$

I \_\_\_\_\_ 2

Prediction (E . (5.6)):

$$k_{ta} = 50 - 10u_A = 30$$

$$k_{ta}\Delta u = e_{PA} \quad \Delta u = \frac{1}{30}20 = 0.6667 \quad u_B = u_A + \Delta u = 2.6667$$

Correction (E . (5.7)):

$$k = 50 - 5u_B = 36.6667$$

$$e_{PB} = P_1 \quad ku_B = 2.2222$$

Q: \_\_\_\_\_ ?

A:

MATLAB

:

```
%
% a single nonlinear spring (Newton-Raphson method)
%
u = 0;
P = 100;
e = P;
conv = e ^ 2 / P ^ 2;
tol = 1.0e-12;
iter = 0;
while conv > tol && iter < 20
    iter = iter + 1;
    uold = u;
    kt = 50 - 10*uold;
    delta_u = kt\e;
    u = uold + delta_u;
    k = 50 - 5*u;
    e = P - k*u;
    conv = e ^ 2 / P ^ 2;
    fprintf('%3d %10.5f %7.5e \n',iter, e, conv);
end
```

T \_\_\_\_\_ :

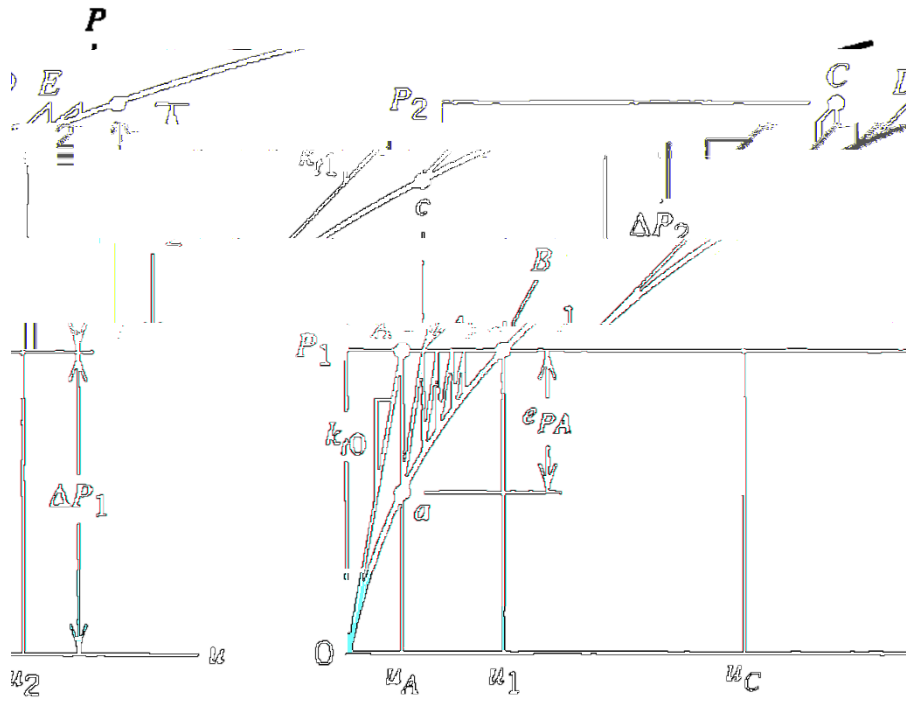
1	20.00000	4.00000e-02
2	2.22222	4.93827e-04
3	0.04535	2.05676e-07
4	0.00002	4.21494e-14

I 4

$$\frac{\|e_P\|}{\|P\|}$$

$$1 \times 10^{-12}$$

6.3.2 Modified Newton-Raphson Method for Single DOF



$$\begin{matrix}
 T & & N & & -R & & & & N & & -R \\
 & & & & & & k_t & & & & \\
 \Delta u, & & & & & & & & & & .T \\
 & & & & & & & & k_{t0} & & \\
 P_1 & & & k_{t1} & & & k_{t1} & & & & P_2.
 \end{matrix}$$

MATLAB :

```

%
% a single nonlinear spring (modified Newton-Raphson method)
%
u = 0;
P = 100;
e = P;
conv = e ^ 2 / P ^ 2;
tol = 1.0e-12;
iter = 0;
while conv > tol && iter < 30
    iter = iter + 1;
    uold = u;
    kt = 50;
    delta_u = kt\e;
    u = uold + delta_u;
    k = 50 - 5*u;
    e = P - k*u;

```



S  $\mathbf{u}^i$  T  $\mathbf{u}^i$  T :

$$(5.10) \quad \Psi(\mathbf{u}^{i+1}) \approx \Psi(\mathbf{u}^i) + \mathbf{K}_T^i(\mathbf{u}^i) \cdot \Delta \mathbf{u}^i$$

$$(5.11) \quad \mathbf{K}_T^i(\mathbf{u}^i) \equiv \left( \frac{\partial \Psi}{\partial \mathbf{u}} \right)^i$$

T  $\Delta \mathbf{u}^i$   $\mathbf{u}^{i+1}$  C E .(5.10 ) :

$$(5.12) \quad \mathbf{K}_T^i \Delta \mathbf{u}^i = \mathbf{f} - \Psi(\mathbf{u}^i)$$

Remark: E (5.12) , :

(1)  $\mathbf{K}_T^i(\mathbf{u}^i)$  ,  $\mathbf{u}^i$  ;

(2)  $\Delta \mathbf{u}^i$  ,  $\mathbf{u}$  ;

(3) - ,  $\mathbf{u}^i$  ,  $\Delta \mathbf{u}^i$  ,  $\mathbf{u}^{i+1}$  . A :

$$(5.13) \quad \mathbf{u}^{i+1} = \mathbf{u}^i + \Delta \mathbf{u}^i$$

I , :

$$(5.14) \quad \mathbf{R}^{i+1} = \mathbf{f} - \Psi(\mathbf{u}^{i+1})$$

I ,  $\mathbf{u}^{i+1}$  ,  $\mathbf{R}^{i+1}$  . O ,

E \_\_\_\_\_ :  $\mathbf{K}(\mathbf{u})\mathbf{u} = \mathbf{f}$

$$\mathbf{K}_T = \left( \frac{\partial \Psi}{\partial \mathbf{u}} \right) = \left( \frac{\partial (\mathbf{K}\mathbf{u})}{\partial \mathbf{u}} \right) = \mathbf{K} + \frac{\partial \mathbf{K}}{\partial \mathbf{u}} \mathbf{u}$$

T  $\mathbf{K}$   $\mathbf{u}$   
 DOF. F , 2 :

$$\begin{bmatrix} K & K \\ K & K \end{bmatrix} \begin{bmatrix} u \\ u \end{bmatrix} = \begin{bmatrix} f \\ f \end{bmatrix}$$

L

$$\Psi_1(u_1, u_2) = K_{11}u_1 + K_{12}u_2$$

$$\Psi_2(u_1, u_2) = K_{21}u_1 + K_{22}u_2$$

Q: E  $\cdot$  (hint:  $\mathbf{K}_T = \left( \frac{\partial \Psi}{\partial \mathbf{u}} \right) = \left( \frac{\partial (\mathbf{K}\mathbf{u})}{\partial \mathbf{u}} \right) = \mathbf{K} + \frac{\partial \mathbf{K}}{\partial \mathbf{u}} \mathbf{u}$ )

A:

F :  $n$  ,

$$\mathbf{K}_T = \mathbf{K} + \frac{\partial \mathbf{K}}{\partial \mathbf{u}} \mathbf{u} \Rightarrow (K_T)_{ij} = K_{ij} + \sum_{m=1}^n \frac{\partial K_{im}}{\partial u_j} u_m$$

**Remarks:**

:

1. A ( )  
 . A .
2. T . T  
 . H ,  
 ,  
 . S , N -R ,  
 .