

Spring 2021

控制系統
Control Systems

Unit 3E
Effects of Zeros and Additional Poles

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- Same Poles, Different Zeros

$$\begin{aligned}
 H_1(s) &= \frac{2}{(s+1)(s+2)} \\
 &= \frac{2}{s+1} - \frac{2}{s+2}
 \end{aligned}$$

$$\begin{aligned}
 H_2(s) &= \frac{2(s+1.1)}{1.1(s+1)(s+2)} \\
 &= \frac{2}{1.1} \left(\frac{0.1}{s+1} + \frac{0.9}{s+2} \right) \\
 &= \frac{0.18}{s+1} + \frac{1.64}{s+2}
 \end{aligned}$$

- One zero at $z = -1.1$ cancels the effect of the pole at $p = -1$

- Same Poles, Different Zeros

$$H(s) = \frac{\frac{s}{\alpha\zeta\omega_n} + 1}{\left(\frac{s}{\omega_n}\right)^2 + 2\zeta\left(\frac{s}{\omega_n}\right) + 1}$$

- Zero: $s = -\alpha\zeta\omega_n = -\alpha\sigma$

- If $\alpha \gg 1$,

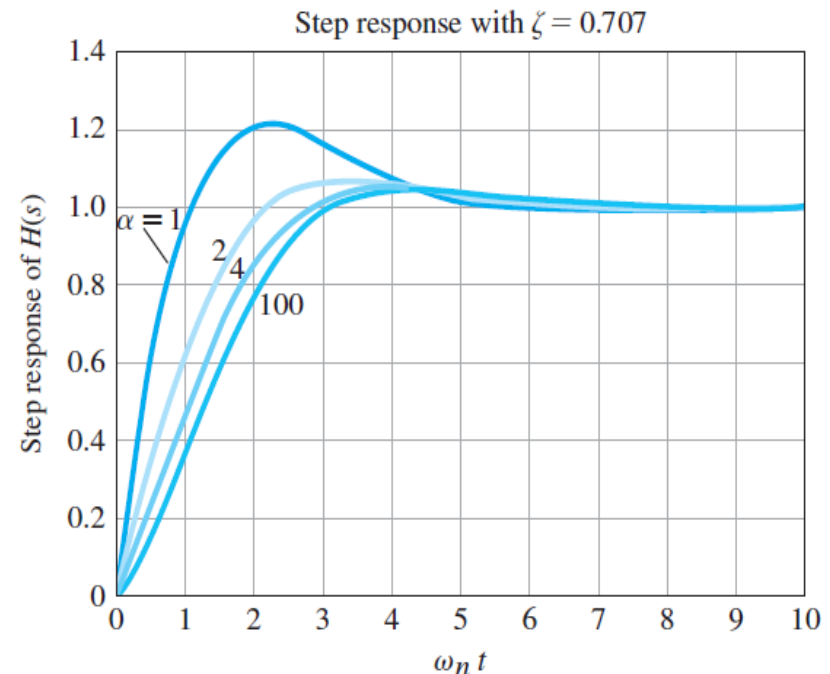
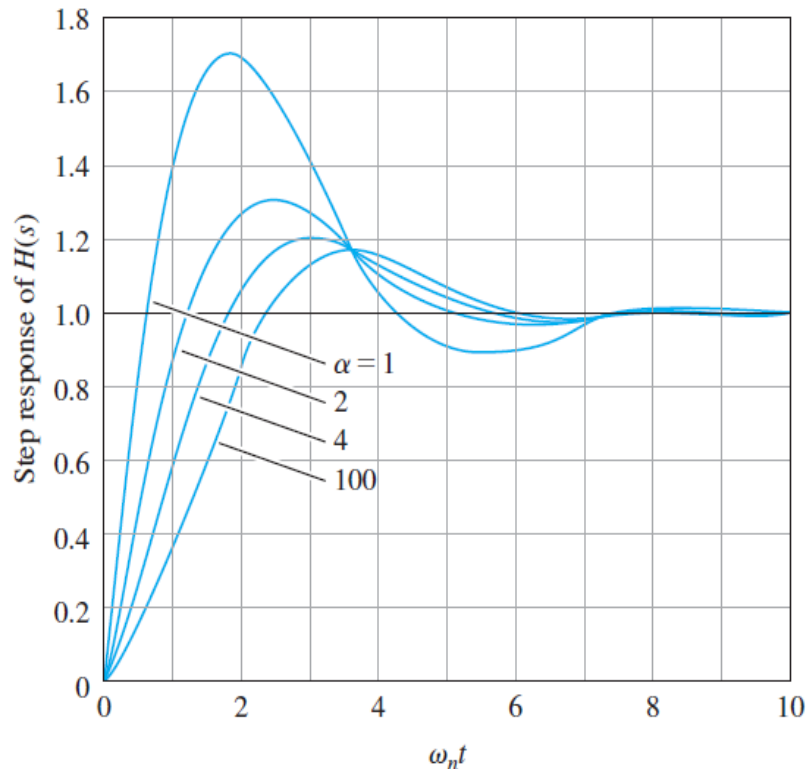
the zero will be far removed from the poles and the zero will have little effect on the response.

- If $\alpha = 1$,

the zero will have a substantial influence on the response.

Same Poles, Different Zeros

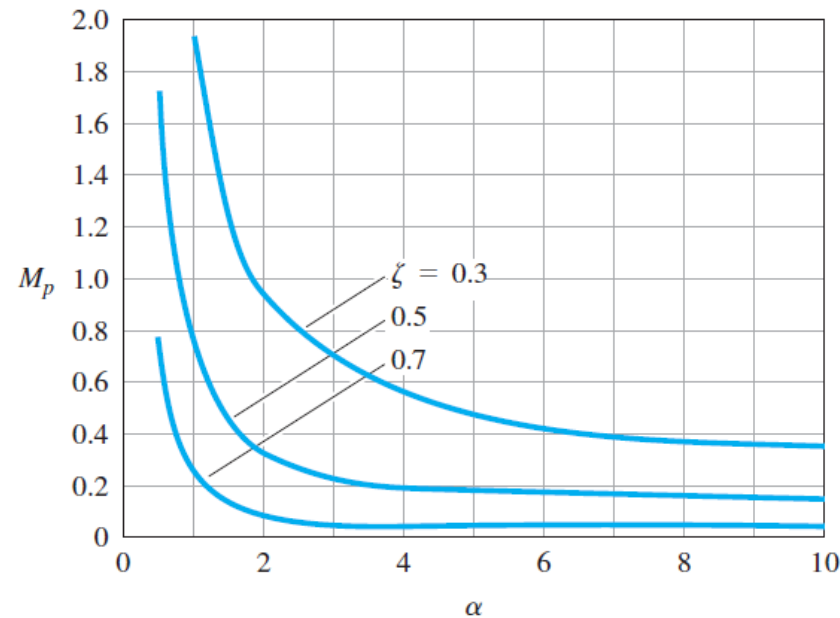
- Plots of the **step response** of a second-order system with a zero ($\zeta = 0.5$)
- Plots of the **step response** of a second-order system with a zero ($\zeta = 0.707$)



- Increase **Overhoot** M_p and reduce **Rise Time** t_r
- Little influence on **Settling Time** t_s

Same Poles, Different Zeros

- Plot of **Overshoot** M_p as a function of **normalized zero location** α .
At $\alpha = 1$, the real part of the zero **equals** the real part of the poles



- Same Poles, Different Zeros

$$H(s) = \frac{\frac{s}{\alpha\zeta w_n} + 1}{\left(\frac{s}{w_n}\right)^2 + 2\zeta\left(\frac{s}{w_n}\right) + 1}$$

- By normalizing frequency

$$\Rightarrow H(s) = \frac{\frac{s}{\alpha\zeta} + 1}{s^2 + 2\zeta s + 1}$$

$$\tau \triangleq w_n t$$

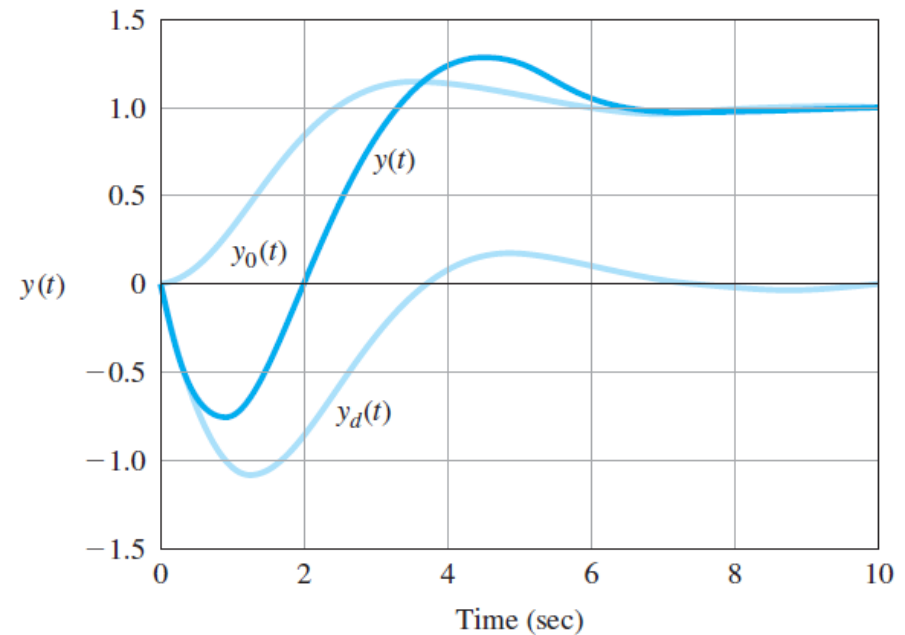
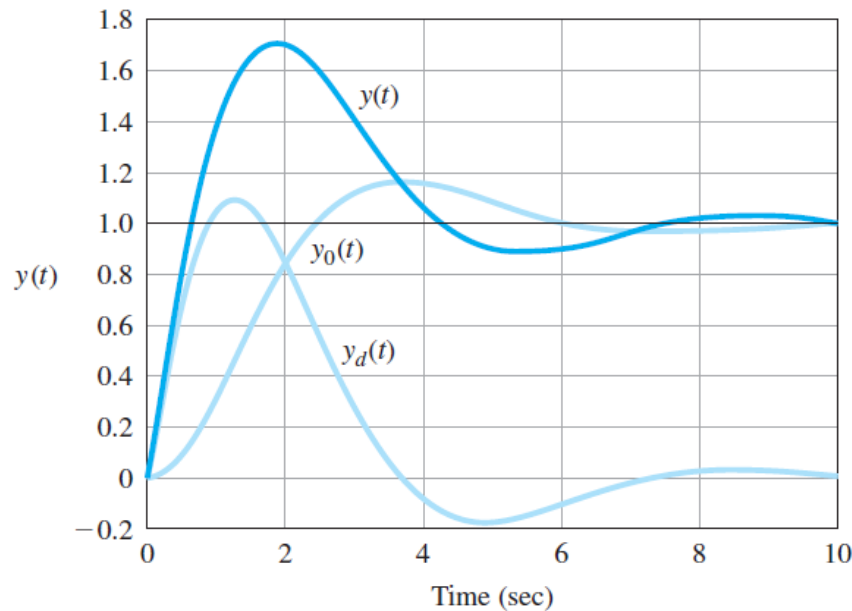
$$= \frac{1}{s^2 + 2\zeta s + 1} + \left(\frac{1}{\alpha\zeta}\right) \left(\frac{s}{s^2 + 2\zeta s + 1}\right)$$

$$\triangleq H_0(s) + H_d(s)$$

$$\Rightarrow y(t) = y_0(t) + y_d(t) = y_0(t) + \frac{1}{\alpha\zeta} \dot{y}_0(t)$$

Effects of Zeros

- Same Poles, Different Zeros
- Second-order step responses $y(t)$ of the transfer functions $H(s)$, $H_0(s)$, and $H_d(s)$
- Step responses $y(t)$ of a second-order system with a zero in the RHP: a nonminimum-phase system



- Zero of $H_d(s)$ increase **Overshoot M_p**

- Example 3.28: Effect of Proximity of Zero to Pole Locations on the Transient Response

$$H(s) = \frac{24}{z} \frac{s + z}{(s + 4)(s + 6)} \quad z = \{1, 2, 3, 4, 5, 6\}$$

$$\begin{aligned} Y(s) &= H(s) \frac{1}{s} = \frac{24}{z} \frac{s + z}{s(s + 4)(s + 6)} \\ &= \frac{24}{z} \frac{s}{s(s + 4)(s + 6)} + \frac{24}{s(s + 4)(s + 6)} \end{aligned}$$

$$y(t) = y_1(t) + y_2(t)$$

$$y_1(t) = \frac{12}{z} e^{-4t} - \frac{12}{z} e^{-6t}$$

$$y_2(t) = z \int_0^t y_1(\tau) d\tau = -3 e^{-4t} + 2 e^{-6t} + 1$$

$$y(t) = 1 + \left(\frac{12}{z} - 3 \right) e^{-4t} + \left(2 - \frac{12}{z} \right) e^{-6t}$$

Example 3.28: Effect of Proximity of Zero to Pole Locations on the Transient Response

Effect of zero on transient response

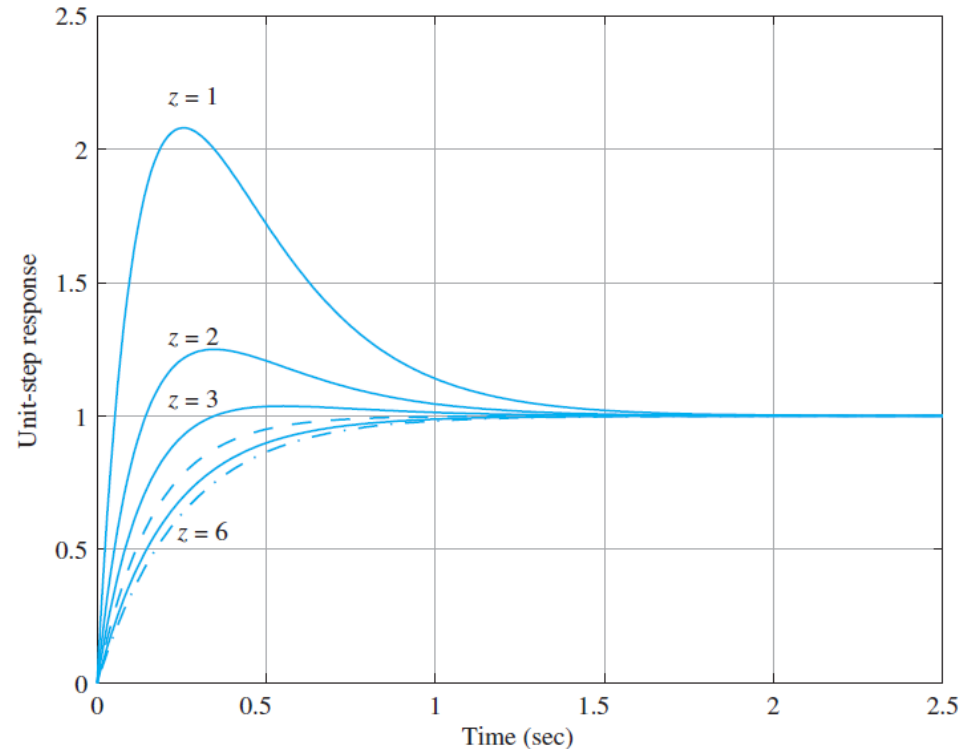
```

z=1;
sys1 = 4*6*(1/z*s+1)/((s+4)*(s+6));
[y1] = step(sys1,t);
plot(t,y1,'LineWidth',2);
hold on;

...

z=6;
sys6 = 4*6*(1/z*s+1)/((s+4)*(s+6));
[y6] = step(sys6,t);
plot(t,y6,'-','LineWidth',2);

```



- Influence of **zero** on response **overshoot**
- **z = 4** or **z = 6**: **absent** due to **zero-pole cancelations**
- **z = 5**: **no overshoot**

Effects of Zeros

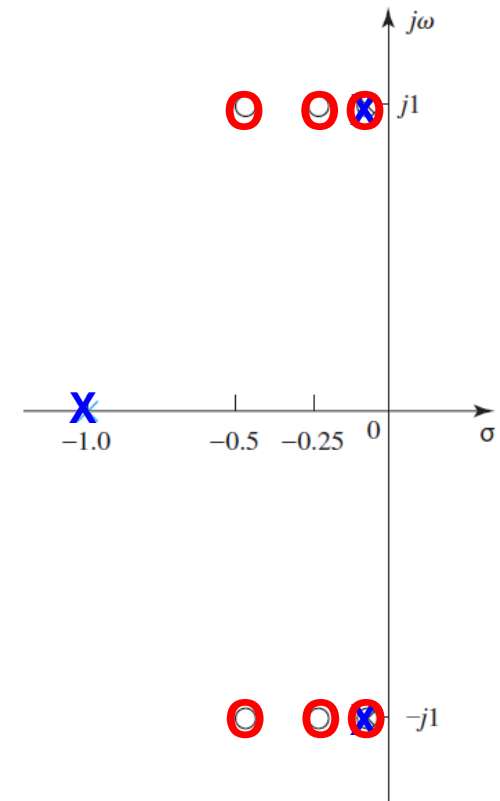
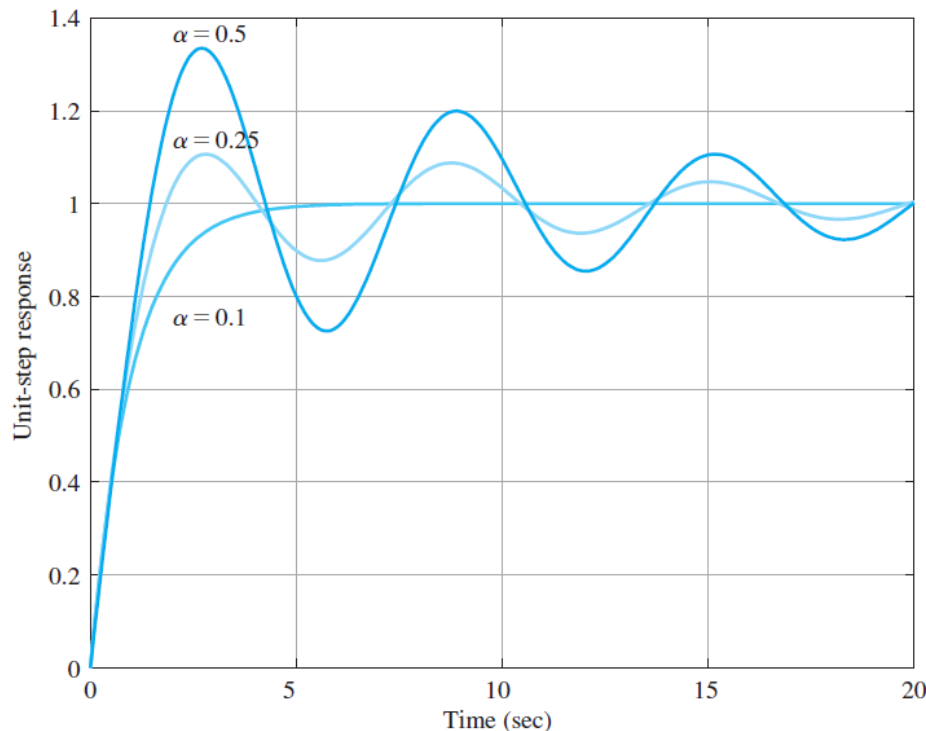
Example 3.29: Effect of Proximity of Complex Zeros to Lightly Damped Poles

$$H(s) = \frac{(s + \alpha)^2 + \beta^2}{(s + 1) [(s + 0.1)^2 + 1]}$$

$$s = -\alpha + j\beta$$

● Locations of complex zeros

$$(\alpha, \beta) = (0.1, 1.0), (0.25, 1.0), (0.5, 1.0)$$



● Effect of complex zeros on transient response

Example 3.30: Aircraft Response Using Matlab

$$\frac{h(s)}{\delta_e(s)} = \frac{30(s-6)}{s(s^2+4s+13)}$$

```
s = tf('s')
```

```
u = -1;
```

```
sysG = u*30*(s-6)/(s^3 + 4*s^2 + 13*s);
```

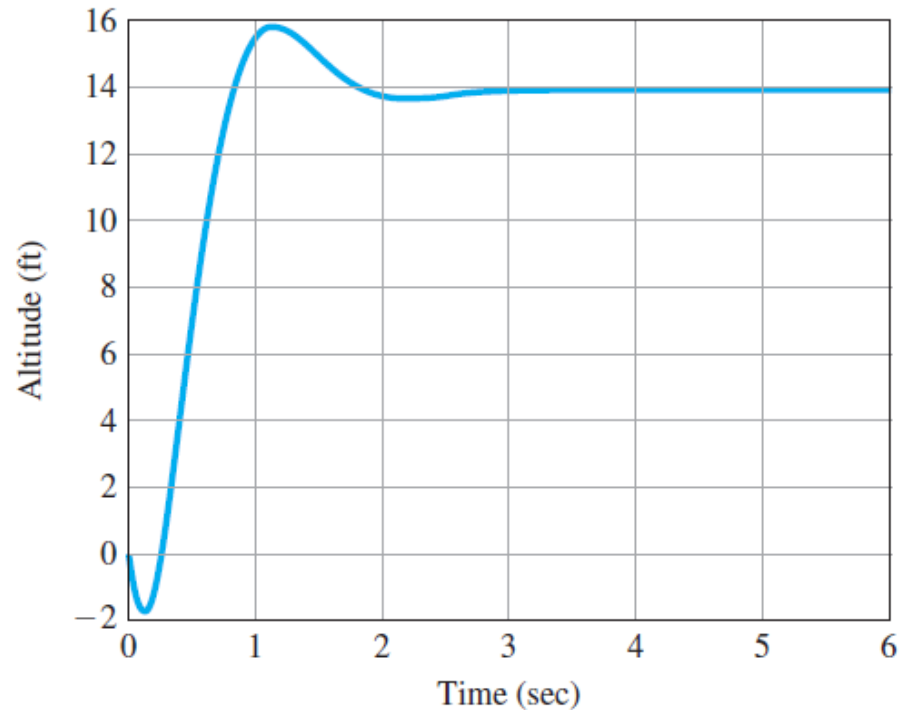
```
t = 0:0.1:6;
```

```
y = impulse(sysG, t);
```

```
plot(t, y)
```

```
grid
```

```
hold on;
```



Final Value:

$$s \frac{30(s-6)(-1)}{s(s^2+4s+13)} \Big|_{s=0} = \frac{30(-6)(-1)}{13} = 13.8$$

- Example 3.30: Aircraft Response Using Matlab

- Rise Time t_r

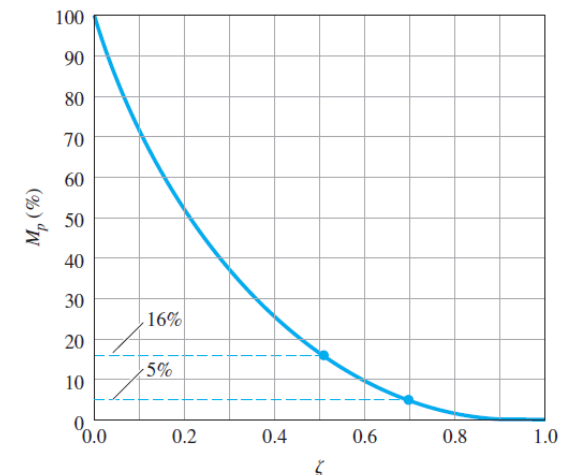
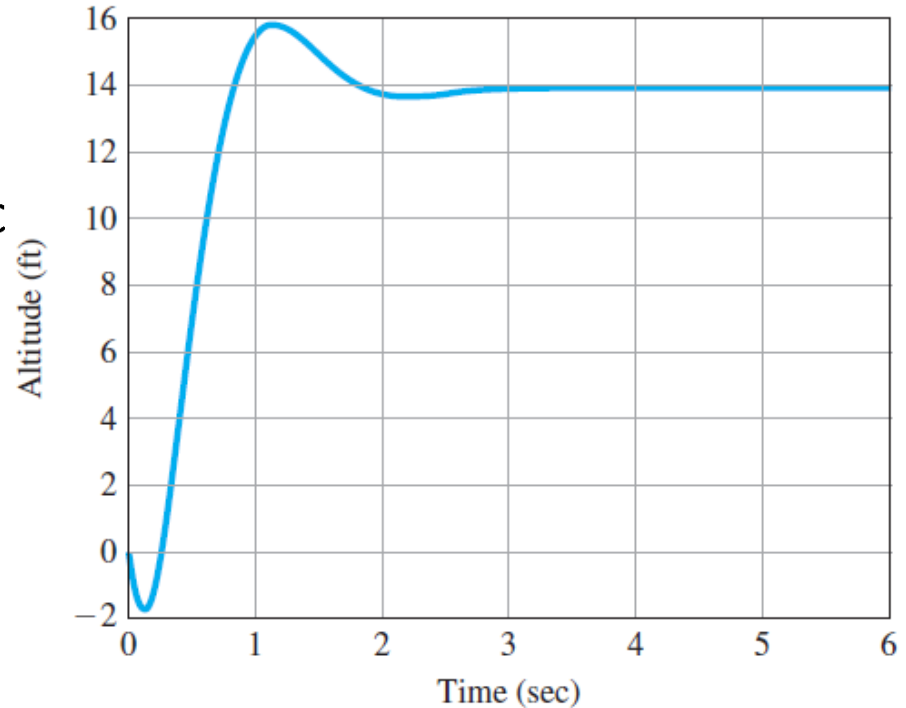
$$t_r \approx \frac{1.8}{\omega_n} = \frac{1.8}{\sqrt{13}} = 0.5 \text{ sec}$$

$$2\zeta\omega_n = 4$$

$$\zeta = \frac{2}{\sqrt{13}} = 0.55$$

$$\Rightarrow M_p = 14\% = 0.14$$

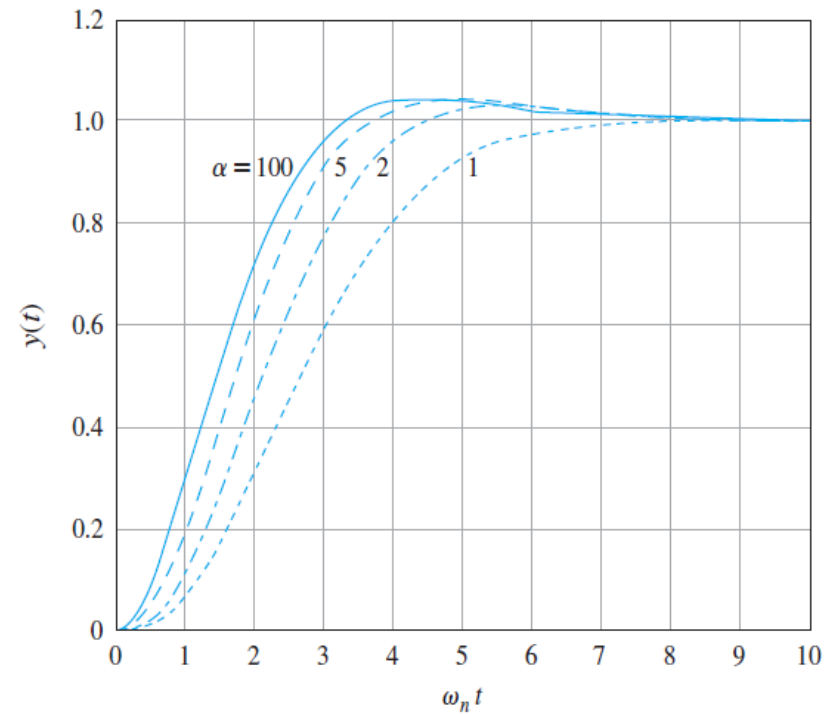
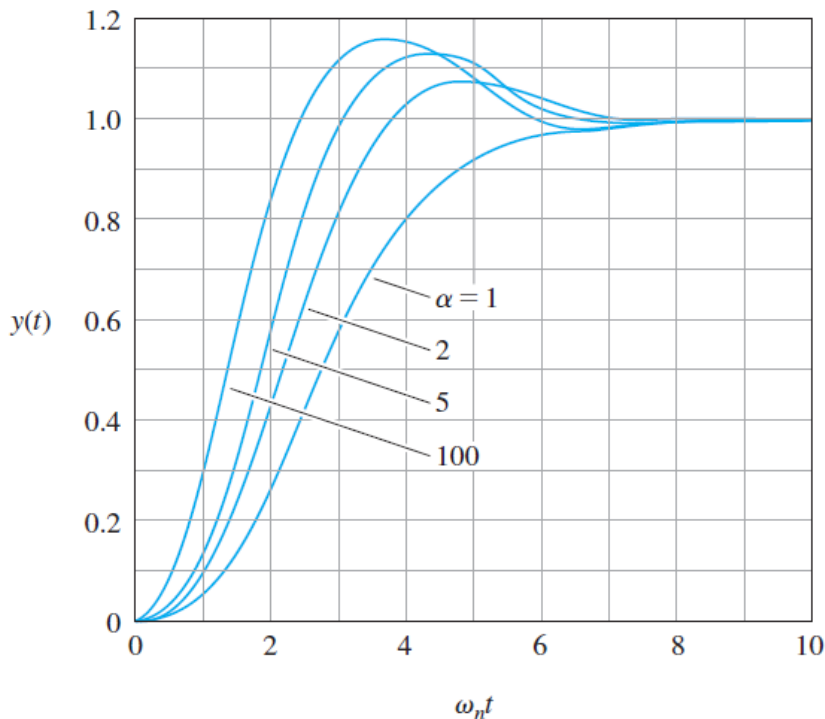
$$\Rightarrow t_s = \frac{4.6}{\zeta\omega_n} = \frac{4.6}{\sigma} = \frac{4.6}{2} = 2.3 \text{ sec}$$



Effects of Pole-Zero Patterns on Dynamic Response

$$H(s) = \frac{1}{\left(\frac{s}{\alpha\zeta\omega_n} + 1\right)\left[\left(\frac{s}{\omega_n}\right)^2 + 2\zeta\left(\frac{s}{\omega_n}\right) + 1\right]}$$

- Step responses for several third-order systems with $\zeta = 0.5$
- Step responses for several third-order systems with $\zeta = 0.707$

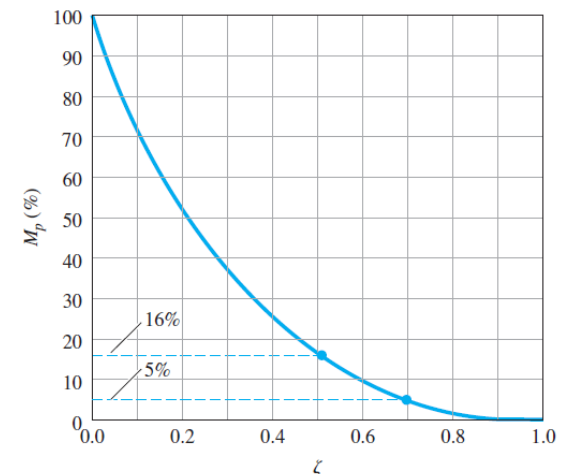


Effects of Pole-Zero Patterns on Dynamic Response

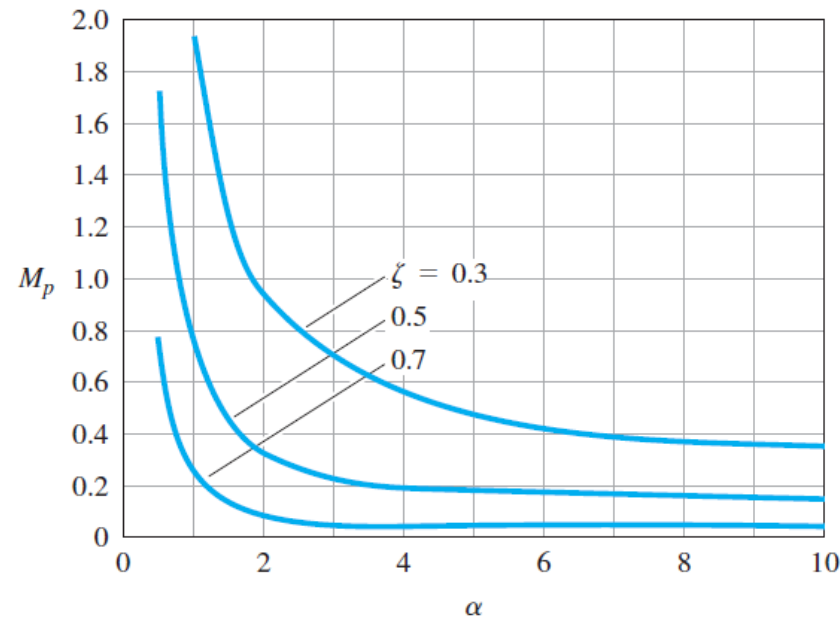
▪ Rise time t_r $\Rightarrow t_r \approx \frac{1.8}{\omega_n}$

▪ Overshoot M_p $\Rightarrow M_p = \begin{cases} 5\%, & \zeta = 0.7 \\ 16\%, & \zeta = 0.5 \\ 35\%, & \zeta = 0.3 \end{cases}$

▪ Settling time t_s $\Rightarrow t_s = \frac{4.6}{\zeta\omega_n} = \frac{4.6}{\sigma}$



- Effects of Pole-Zero Patterns on Dynamic Response
- A zero in LHP will increase the overshoot if the zero is within a factor of 4 of the real part of the complex poles.
- A zero in RHP will depress the overshoot.



- Effects of Pole-Zero Patterns on Dynamic Response

- An additional pole in the LHP

will increase the rise time significantly

if the extra pole is within a factor of 4

of the real part of the complex poles.

- Normalized rise time for several locations of an additional pole

