

Spring 2020

控制系統
Control Systems

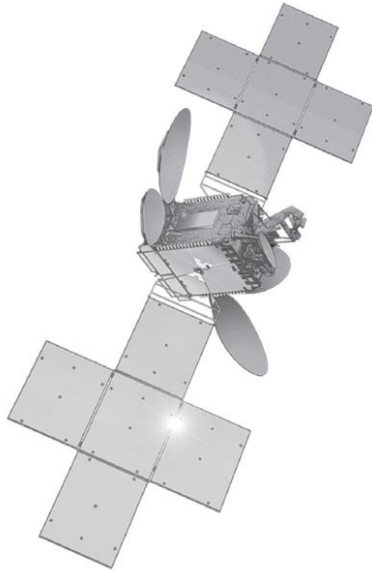
Unit 22
Mechanical Systems – Rotational Motion

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NTU-EE

Mar 2020 – Jul 2020

- Communication satellite



Source: Courtesy Thaicom PLC and Space Systems/Loral

- The purpose is to control the **attitude** of the satellite, such as
 - ✓ **Antennas** point toward earth
 - ✓ **Solar panels** orient toward the sun

■ Model (Equations of Motion: Rotational motion)

$$M = I \alpha$$

- M ($N \cdot m^2$): the sum of all external moments about the center of mass,
- I ($Kg \cdot m^2$): the body's mass moment of inertia about its center of mass,
- α (rad/sec^2): the angular acceleration of the body

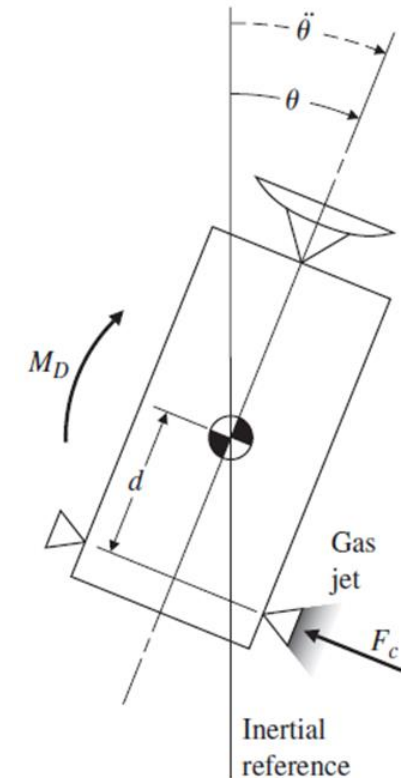
Example 2.3 (Rotational motion): Satellite Attitude Control Model

■ Model (Equations of Motion)

- Three axes, consider one axis at a time

$$F_c \cdot d + M_D = I \cdot \ddot{\theta}$$

- $F_c \cdot d$: Moments of control force
- M_D : Moments of small disturbance



■ Transfer Function

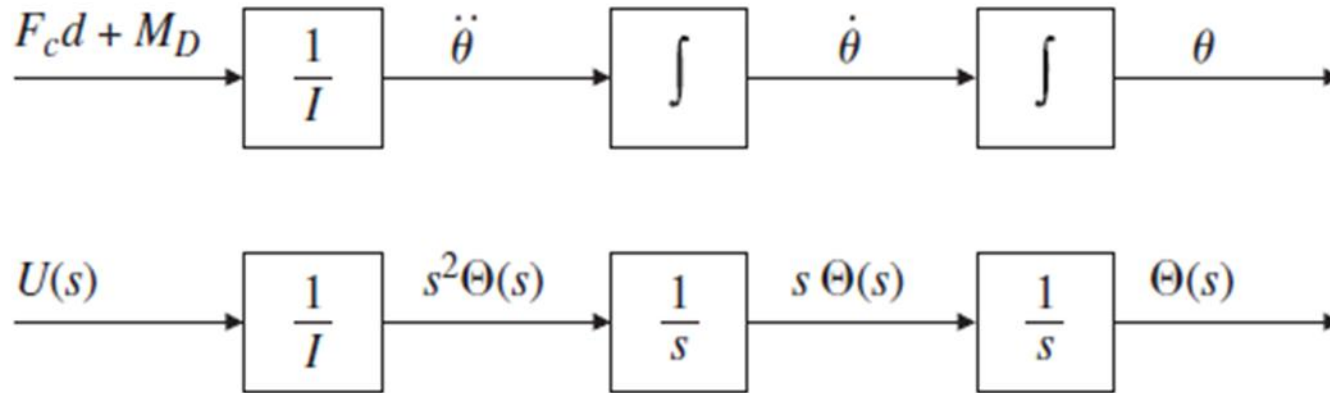
- Let $F_c \cdot d + M_D = u$

$$\frac{\Theta(s)}{U(s)} = \frac{1}{I} \cdot \frac{1}{s^2} \quad (\text{Double-Integrator plant})$$

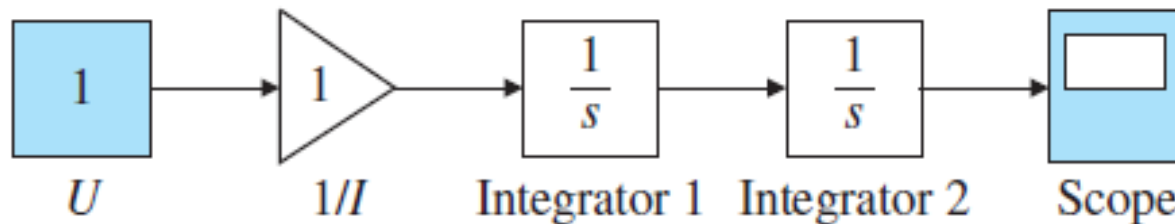
Example 2.3 (Rotational motion): Satellite Attitude Control Model

- Block diagram

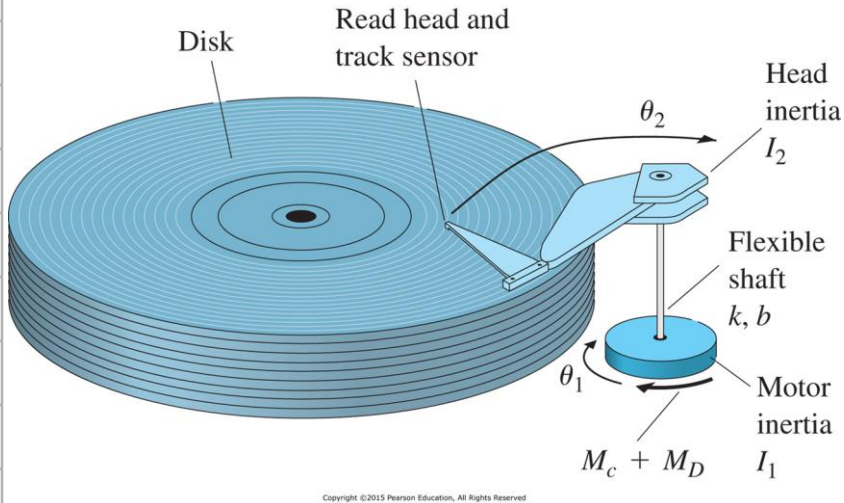
$$\frac{\Theta(s)}{U(s)} = \frac{1}{I} \cdot \frac{1}{s^2} \quad (\text{Double-Integrator plant})$$



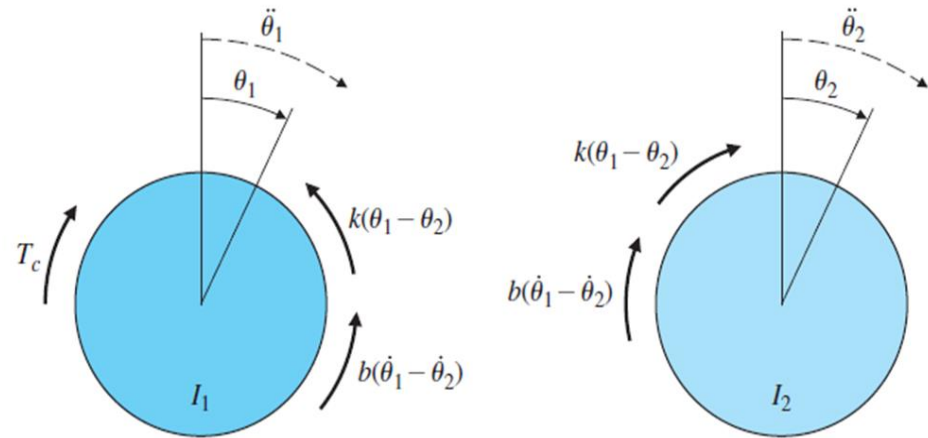
- Simulink



- Disk read/write head



- The moment of each body: free body diagram



- Model (Equations of Motion: Rotational motion)

$$I_1 \ddot{\theta}_1 + b(\dot{\theta}_1 - \dot{\theta}_2) + k(\theta_1 - \theta_2) = M_c + M_D$$

$$I_2 \ddot{\theta}_2 + b(\dot{\theta}_2 - \dot{\theta}_1) + k(\theta_2 - \theta_1) = 0$$

- M_c : Moments of applied control
- M_D : Moments of small disturbance

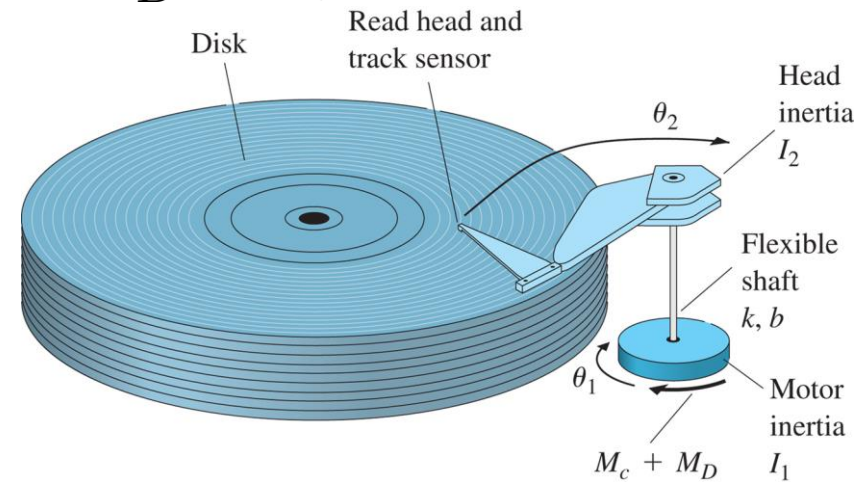
Example 2.4 Flexible Read/Write for a Disk Drive

Model (Equations of Motion)

- Simplify the model, consider the case $M_D = 0, b = 0$

$$I_1 \ddot{\theta}_1 + k(\theta_1 - \theta_2) = M_c$$

$$I_2 \ddot{\theta}_2 + k(\theta_2 - \theta_1) = 0$$



Transfer Function

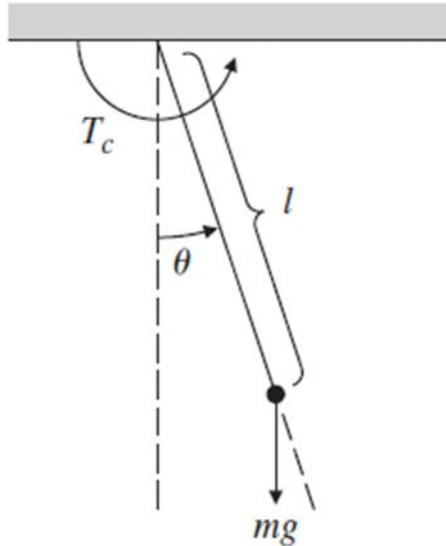
$$\frac{\Theta_2(s)}{M_c(s)} = \frac{k}{I_1 I_2 s^2 (s^2 + \frac{k}{I_1} + \frac{k}{I_2})}$$

$$\frac{\Theta_1(s)}{M_c(s)} = \frac{I_2 s^2 + k}{I_1 I_2 s^2 (s^2 + \frac{k}{I_1} + \frac{k}{I_2})}$$

- “**Noncollocated case**”: there is **flexibility** between the sensor and the actuator

- “**Collocated case**”: the sensor and the actuator are **rigidly attached** to one another

- Pendulum



- Model (Equations of Motion)

$$T_c - mgl \sin \theta = I\ddot{\theta}$$

- The moments of inertia about the pivot point is $I = ml^2$

$$\ddot{\theta} + \frac{g}{l} \sin \theta = \frac{T_c}{ml^2}$$

- The model is nonlinear due to $\sin \theta$
- When the motion is small, i.e., θ small, $\sin \theta \approx \theta$

$$\ddot{\theta} + \frac{g}{l} \theta = \frac{T_c}{ml^2} \quad (\text{Linearization model})$$

Example 2.5 Pendulum

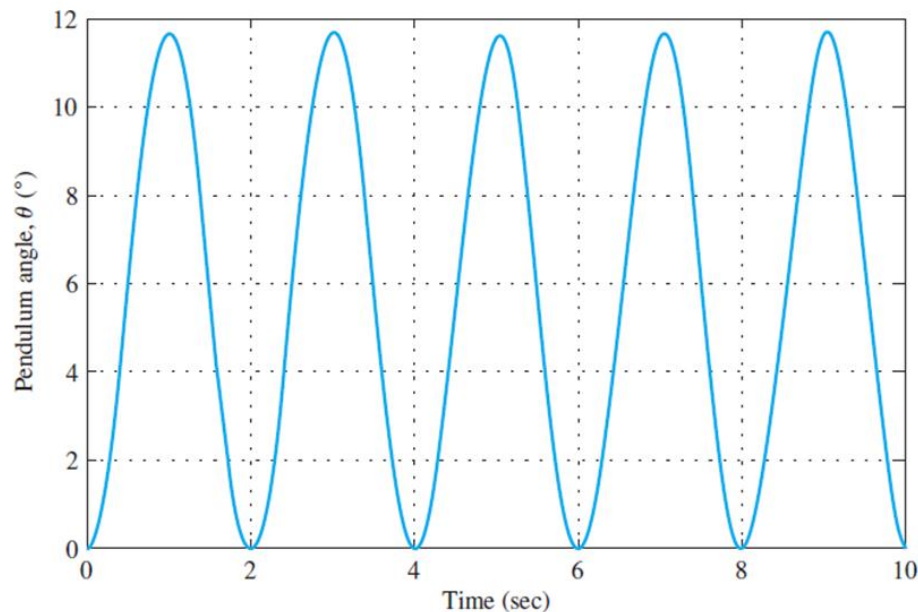
Transfer Function

$$\frac{\Theta(s)}{T_c(s)} = \frac{1}{s^2 + \frac{g}{l}}$$

Matlab code

- `t=0:0.02:10;`
- `m=1; L=1; g=9.81;`
- `s = tf('s');`
- `sys = (1/(m*L^2))/(s^2+g/L) ;`
- `Y=step(sys,t);`

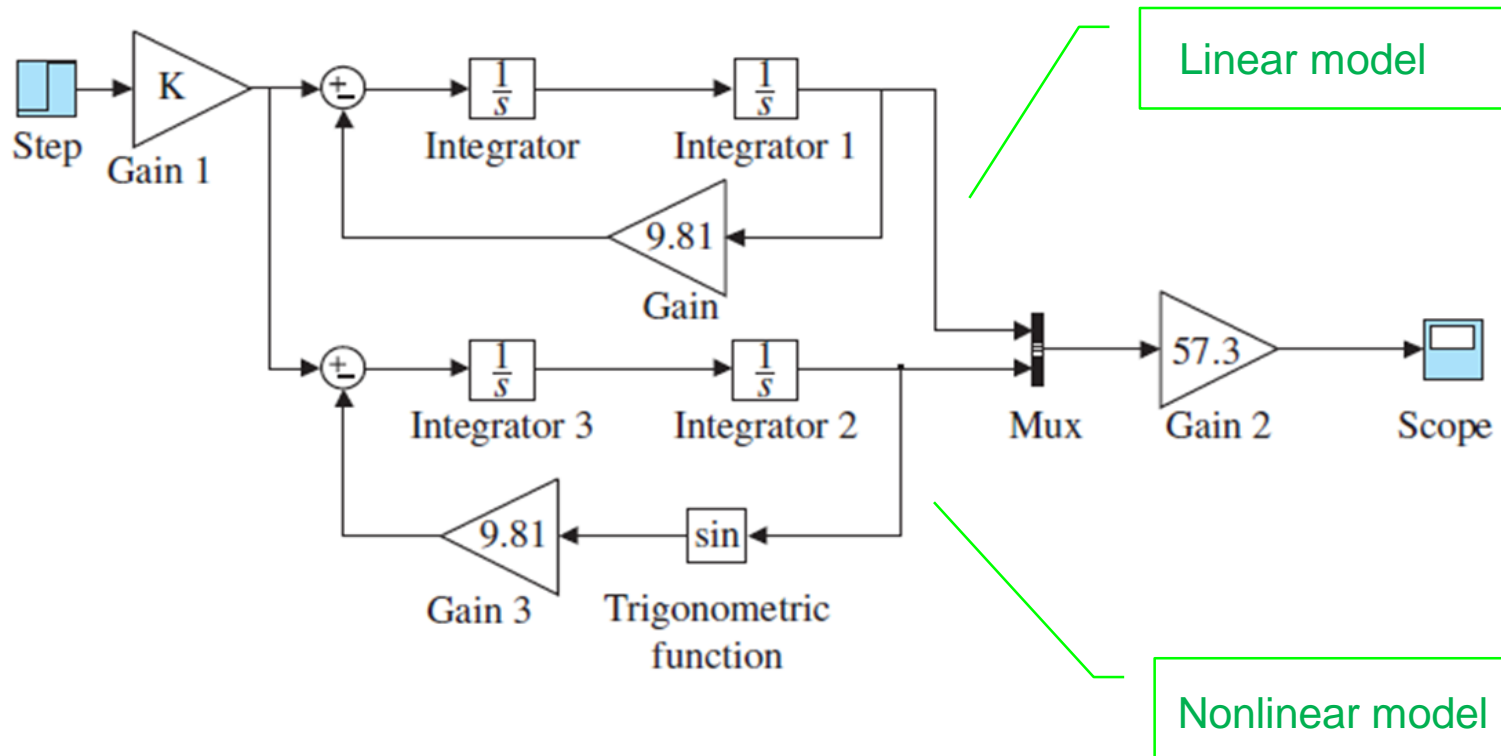
- `Rad2Deg=57.3;`
- `Plot(t,Rad2Deg*y) %converts output from radians to degrees`



Example 2.6 Pendulum (Simulink for nonlinear motion)

- Matlab Simulink (m=1; L=1; g=9.81)

$$\ddot{\theta} + \frac{g}{l}\theta = \frac{T_c}{ml^2}$$

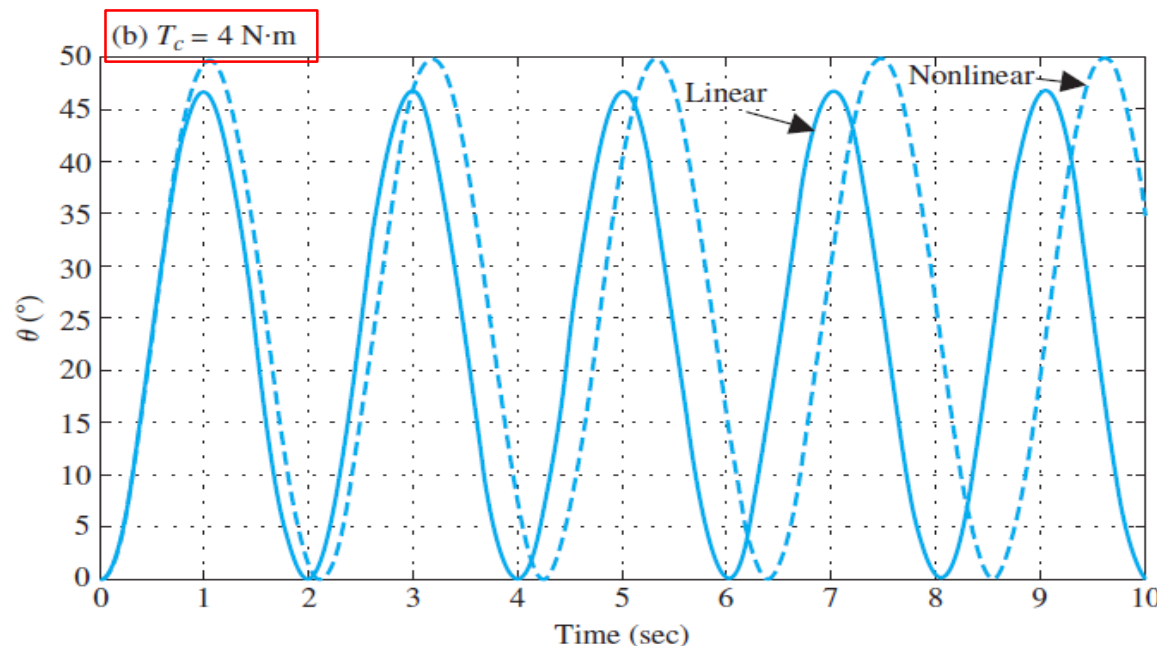
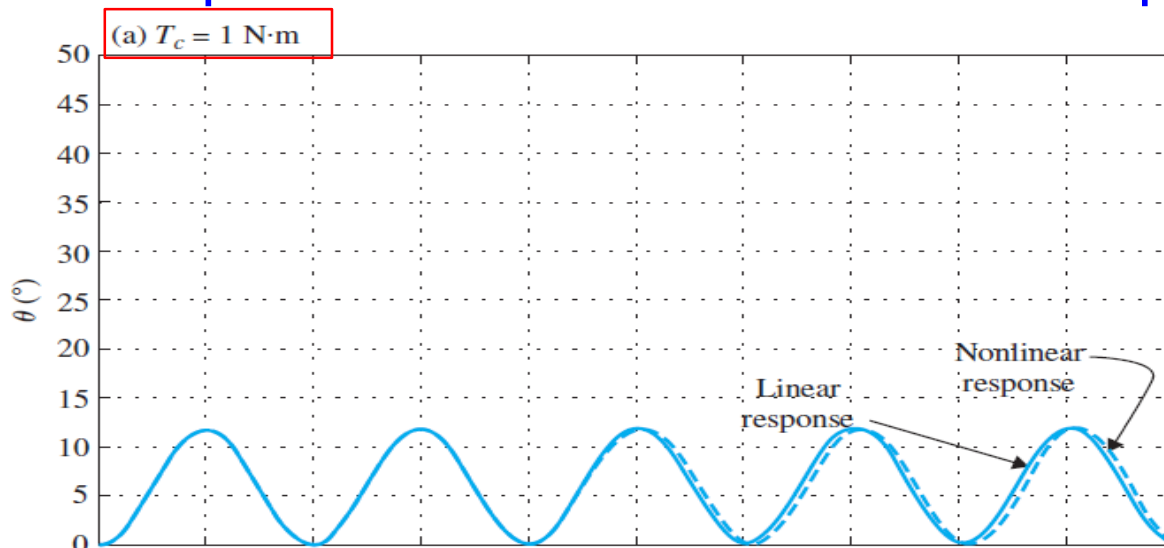


Linear model

Nonlinear model

$$\ddot{\theta} + \frac{g}{l}\sin\theta = \frac{T_c}{ml^2}$$

Comparisons of linear & nonlinear responses



$$\ddot{\theta} + \frac{g}{l} \sin \theta = \frac{T_c}{ml^2}$$

$$\ddot{\theta} + \frac{g}{l} \theta = \frac{T_c}{ml^2}$$

- When $T_c=1$, the output θ remains small, thus the approximation is still good ($\sin \theta \approx \theta$)
- When $T_c=4$, the output θ becomes large, thus the approximation is not good ($\sin \theta \approx \theta$ does not hold)